Preactuated Multirate Feedforward Control for Independent Stable Inversion of Unstable Intrinsic and Discretization Zeros

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Abstract—A plant with unstable zeros is known to be difficult to be controlled, because of the initial undershoot of its step response and the unstable poles in its inversion system. There are two reasons why a plant has unstable zeros in the discrete-time domain: 1) noncollocation of actuators and sensors; and 2) discretization by the zeroorder hold. Addressing problems 1) and 2) simultaneously, approximate and discrete-time stable inversion methods have been proposed as model-based feedforward control for unstable zeros. However, the approximated inversion methods compromise tracking performance, and the discrete-time stable inversion methods obtain oscillatory control inputs owing to the direct cancellation of discretization zeros. Therefore, this paper proposes a preactuated multirate feedforward control, which is an independent stable inversion of the two types of zeros: unstable intrinsic zeros from noncollocation and unstable discretization zeros from the zero-order hold. The proposed method combines both a state trajectory generation based on a time- and imaginary-axis reversal to address problem 1) and a multirate feedforward implementation to address problem 2), independently. Moreover, this study verifies the tracking performance improvement by performing simulations and experiments in comparison with well-known model-based feedforward control methods using a high-precision motion stage. The role and importance of multirate feedforward control are also demonstrated by comparing it to a continuous-time domain approach with preactuation.

Index Terms—Discretization, multirate feedforward, nonminimum phase (NMP) system, preactuation, unstable zero.

I. INTRODUCTION

TRACKING control with zero gain and phase errors between the desired and output trajectories at every

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0.8 0.60.4 $y\left(t
ight)$ 0.2 -1)0 (s+1) $(s+1)^{3}$ -0.2 (s+1)-0.4 0 2 8 4 6 10 Time [s]

Fig. 1. Step response comparison: P_1 is a first-order transfer function without an unstable zero. P_2 , P_3 , and P_4 have one, two, and three unstable zero(s), respectively.

sampling point can be achieved, in theory, by model inversion (i.e., pole-zero cancellation). However, a plant with unstable (i.e., nonminimum phase, NMP) zeros gives rise to an initial undershoot when stepped (see Fig. 1) and highly oscillatory or unstable control trajectories when inverted [1], [2]. Unstable zeros of discrete transfer functions can be classified as: 1) intrinsic zeros corresponding to the plant dynamics (e.g., noncollocated placement of sensor and actuator); or 2) discretization zeros due to signal sampling [3]. Note that discretization zeros are unstable when the relative order of the continuous-time plant is greater than two, even without continuous-time unstable zeros [4]. Systems with unstable zeros are, for instance, wafer stages [5], high-precision positioning stages for flat panel display manufacturing systems [6], [7], hard disk drives [8], boost converters [9], permanent magnet synchronous motors during six-step operation [10], backward driving cars [11], and pitch angle regulation of aircraft [12].

Extensive research has been dedicated to the design of tracking control methods for systems with NMP zeros. The existing proposals can be classified as approximate model-inversion methods and preactuated model-inversion methods [13]. A straightforward way to implement an approximate inversion is to cancel stable poles and zeros while ignoring NMP zeros, i.e., NPZI method [14]. This approach exhibits both magnitude

1083-4435 © 2019 IEEE. Personal use is permitted, but republication/redistribution requires IEEE permission. See http://www.ieee.org/publications_standards/publications/rights/index.html for more information. and phase errors; hence, zero-magnitude-error tracking control (ZMETC) [15] and zero-phase-error tracking control (ZPETC) [16] methods have been proposed and improved upon. Despite the stable control input, NPZI, ZMETC, and ZPETC may not yield satisfactory tracking performance owing to the approximations involved (depending on the system and performance specifications).

Preactuated model inversion can achieve perfect tracking with infinite preview (i.e., knowledge of future references) and preactuation (i.e., actuation applied a time interval before the actual output of the system). Continuous-time-domain stable inversion methods are proposed in [17]-[20]. This study unifies these proposals as continuous-time preactuated model-inverse (CPMI) methods. However, these approaches do not consider the effect of the zero-order hold. There lies the key difference between the CPMI method and the proposed preactuated multirate feedforward control, which explicitly considers the effect of the zero-order hold by constructing a multirate system. Alternatively, a discrete-time-domain approach is proposed in [21], which compensates both intrinsic and discretization unstable zeros, simultaneously. This method can achieve perfect tracking at each sampling point, but it can cause high oscillations in the intersampling behavior by direct cancellation of the discretization zeros [1], [22]. For high-precision motion systems such as wafer scanners and printing systems, preactuation methods have been proposed and applied, focusing on iterative learning control and linear periodically time-varying systems [23], [24].

We propose a multirate feedforward with preactuation [25] and experimentally validate its effectiveness [26]. The proposed method solves problems 1) and 2) independently, regardless of the desired trajectory given that it is n-1 differentiable (where n denotes the order of the nominal plant). This paper concludes our findings and provides an in-depth review and comparison of the effectiveness of the independent inversion with a wide range of established techniques in the field.

First, the unstable intrinsic zeros in the continuous-time domain are offset through a state trajectory generation using a time-axis reversal. Then, the stable inversion of the discretization zeros is calculated through a multirate feedforward approach [27]. Note that the multirate control is commonly referred to as the set of techniques used to improve feedback control performance [28]-[30]. These methods use multiple sampling times to enhance the performance/cost tradeoff present in single-rate feedback control. In contrast, this multirate method is for model-based *feedforward* control, using multiples of the sampling time to obtain a stable inversion of the discretization zeros. A multirate *feedforward* method is proposed in [27] and extended to systems with continuous-time stable zeros [7] and adaptive control [31]. Fujimoto et al. [8] proposed a multirate NPZI method, which ignores only intrinsic (continuoustime) unstable zeros, whereas the single-rate NPZI method introduced in Section III-A and [14] ignores both intrinsic and discretization zeros simultaneously. This method [8] constructs a stable inversion of the discretization zeros; however, it is still

affected by the approximation of the unstable intrinsic zeros. The proposal described in this paper goes a step further, improving upon the model inversion and subsequently tracking errors of systems with NMP zeros. It is an independent stable inversion for unstable intrinsic zeros and discretization zeros, which are inverted by preactuation and multirate feedforward, respectively.

To demonstrate the effectiveness, this paper thoroughly compares the proposed method with both approximate and preactuated model-inversion methods by simulations and experiments. This paper shows that it is impossible to compensate for the zero-order hold delay by only previewing the reference of CPMI method. The considered system has several intrinsic and discretization unstable zeros, for which conventional approximate methods are unable to achieve perfect tracking. Simulation and experimental results show that the proposed method effectively reduces tracking errors. Note that the performance of modelbased methods, including the proposed method, is affected by modeling errors, which can be mitigated by feedback control. However, because of the presence of continuous-time unstable zeros in the lower frequency range, it is difficult to design high-bandwidth feedback controllers. Hence, for a system with unstable zeros, precise system identification inherently plays a crucial role. Albeit, for the range of applications targeted, extensive modeling and identification are at present commonly employed.

II. NOTATIONS AND DEFINITIONS

The plant in continuous-time domain is defined as $P_c(s)$. $P_s[z_s]$ denotes the discretized plant of $P_c(s)$ by the zero-order hold with sampling time T_u , where s denotes a complex variable for Laplace transform and $z_s = e^{T_u s}$.

A continuous-time transfer function of the nominal plant is

$$P_n(s) = \frac{B(s)}{A(s)} \tag{1}$$

where A(s) is the *n*th order and B(s) is the *m*th order, expressed as

$$A(s) = \frac{s^{n} + a_{n-1}s^{n-1}\dots + a_{0}}{b_{0}}$$

$$B(s) = \frac{b_{m}s^{m} + b_{m-1}s^{m-1} + \dots + b_{0}}{b_{0}}.$$
(2)

Note that (1) is irreducible. The state and output equations of (1) are defined as

$$\dot{\boldsymbol{x}}(t) = \boldsymbol{A}_c \boldsymbol{x}(t) + \boldsymbol{b}_c u(t), \quad y(t) = \boldsymbol{c}_c \boldsymbol{x}(t). \quad (3)$$

The discretized plant by a zero-order hold with sampling time T_u is defined as

$$\boldsymbol{x}[k+1] = \boldsymbol{A}_s \boldsymbol{x}[k] + \boldsymbol{b}_s \boldsymbol{u}[k], \ \boldsymbol{y}[k] = \boldsymbol{c}_s \boldsymbol{x}[k]$$
(4)

$$\boldsymbol{A}_{s} = e^{\boldsymbol{A}_{c}T_{u}}, \ \boldsymbol{b}_{s} = \int_{0}^{T_{u}} e^{\boldsymbol{A}_{c}\tau} d\tau \cdot \boldsymbol{b}_{c}, \ \boldsymbol{c}_{s} = \boldsymbol{c}_{c}.$$
 (5)



Fig. 2. Approximated plant inverse feedforward control configuration, where $C_{ff}[z_s] = \tilde{P}_n^{-1}[z_s]$. $\tilde{P}_n[z_s]$ is the approximated nominal plant model without unstable zeros. $P_c(s)$ and $C_{fb}[z_s]$ denote the continuous-time plant and the discrete-time feedback controller, respectively. \mathcal{H} and \mathcal{S} denote a holder and a sampler, respectively.



Fig. 3. Example of tracking control performance of single-rate modelinversion methods $C_{ff}[z_s]P_s[z_s]z_s^{n \text{ pre}}$, where n_{pre} denotes the previewed number of samples for ZPETC. The plant shown in Fig. 6(b) is used for the case study.

In the discrete transfer function, it is defined as

$$P_n[z_s] = \boldsymbol{c}_s (z_s \boldsymbol{I} - \boldsymbol{A}_s)^{-1} \boldsymbol{b}_s.$$
(6)

III. SINGLE-RATE MODEL-INVERSION METHODS

A. Approximate Model-Inverse Methods

When a nominal plant $P_n[z_s]$ discretized by the zero-order hold has an unstable zero, the inversion system in the feedforward controller $P_n^{-1}[z_s]$ becomes unstable. To avoid this problem, several approximate model-inverse feedforward controllers $\tilde{P}_n^{-1}[z_s]$ have been proposed in the literature. A generalized block diagram is shown in Fig. 2, in which T_y and T_u denote the sampling and control periods, respectively. In this section, $T_y = T_u$ because only single-rate $(z_s = e^{T_u s})$ control methods are considered. Approximations decompose the nominal plant in a stable $B^{\text{st}}[z_s]$ and unstable part $B^{\text{ust}}[z_s]$, as follows:

$$P_{n}[z_{s}] = \frac{B[z_{s}]}{A[z_{s}]} = \frac{B^{\text{st}}[z_{s}]B^{\text{ust}}[z_{s}]}{A[z_{s}]}$$
(7)

$$B^{\text{ust}}[z_s] = b_{un_u} z_s^{n_u} + b_{u(n_u-1)} z_s^{n_u-1} + \dots + b_{u0}$$
 (8)

where n_u denotes the order of $B^{ust}[z_s]$. The feedforward controller is then designed as

$$C_{ff}[z_s] = \tilde{P}_n^{-1}[z_s] = \frac{z_s^{-q} A[z_s]}{B^{\text{st}}[z_s]\tilde{B}^{\text{ust}}[z_s]}.$$
(9)

The difference among the three prominent methods (NPZI, ZPETC, and ZMETC) is the design of $\tilde{B}^{ust}[z_s]$ and q ($0 \le q \in \mathbb{Z}$). Refer to [14] for an overview. A Bode plot of tracking control performances of single-rate model-inversion methods is shown in Fig. 3, where a plant shown in Fig. 6(b) is used as a case study.

B. CPMI Methods

Several CPMI methods are proposed in [17]–[20]. However, these papers do not consider the effect of the zero-order hold. The key difference between CPMI methods and the proposed method is the zero-order hold consideration by multirate feed-forward control. A block diagram of the CPMI method is shown in Fig. 4.

1) Step 1. Stable–unstable decomposition: $B(s)^{-1}$ defined in (2) is decomposed into a stable part $F^{st}(s)$ and an unstable part $F^{ust}(s)$ as follows:

$$B(s)^{-1} = \frac{b_0}{b_m s^m + b_{m-1} s^{m-1} + \dots + b_0}$$
(10)

$$=F^{\rm st}(s)+F^{\rm ust}(s)\tag{11}$$

$$f^{\rm st}(t) = \mathcal{L}^{-1}\left[F^{\rm st}(s)\right], \bar{f}^{\rm ust}(t) = \mathcal{L}^{-1}\left[F^{\rm ust}(-s)\right]$$
(12)

where \mathcal{L}^{-1} denotes the inverse unilateral Laplace transform. Note that $F^{\text{ust}}(-s)$ is stable.

2) Step 2. Stable part feedforward control input: Stable part feedforward control input is calculated by a convolution between the reference $A_{\text{CPMI}}r_{\text{CPMI}}(t)$ and $f^{\text{st}}(t)$ as

$$u_{ff}^{\rm st}(t) = \int_{-\infty}^{t} f^{\rm st}(t-\tau) \boldsymbol{A}_{\rm CPMI} \boldsymbol{r}_{\rm CPMI}(\tau) d\tau \qquad (13)$$

where

$$\boldsymbol{A}_{\text{CPMI}} = \frac{1}{b_0} \begin{bmatrix} a_0 & a_1 & \cdots & a_{n-1} & 1 \end{bmatrix}$$

$$\boldsymbol{r}_{\text{CPMI}} = \begin{bmatrix} 1 & \rho & \cdots & \rho^n \end{bmatrix}^{\mathrm{T}} \boldsymbol{r}(t)$$
 (14)

where ρ denotes the Heaviside operator [33]. Equation (13) can be written as

$$u_{ff}^{\rm st}(t) = \int_0^t f^{\rm st}(t-\tau) \boldsymbol{A}_{\rm CPMI} \boldsymbol{r}_{\rm CPMI}(\tau) d\tau \qquad (15)$$

assuming $\mathbf{r}_{\text{CPMI}}(t) = \mathbf{0}$ when t < 0.

3) Step 3. Unstable part feedforward control input: Unstable part feedforward control input is calculated by: 1) a convolution between the time-axis reversed reference $A_{\text{CPMI}}r_{\text{CPMI}}(-\bar{\tau})$ and the stable signal $\bar{f}^{\text{ust}}(\bar{t}-\bar{\tau})$; and 2) a time-axis reversal. This procedure results in an infinity time preactuation for



Fig. 4. CPMI method. The stable inversion of the continuous-time plant is calculated through the dotted boxed blocks. Note that this step does not consider the zero-order hold of the plant.



Fig. 5. Proposed preactuated multirate feedforward control. S, H, and L denote a sampler, holder, and lifting operator [32], respectively. z and z_s denote e^{sT_r} and e^{sT_u} , where $T_r = nT_u$, respectively.

continuous-time unstable zeros compensation as

$$u_{ff}^{\rm ust}(t) = \int_{-\infty}^{\bar{t}} \bar{f}^{\rm ust}(\bar{t} - \bar{\tau}) \boldsymbol{A}_{\rm CPMI} \boldsymbol{r}_{\rm CPMI}(-\bar{\tau}) d\bar{\tau} \Big|_{\bar{t} = -t}.$$
 (16)

4) Step 4. Total feedforward control input: The total feedforward control input is calculated by a sum of the stable and the unstable part feedforward control input as

$$u_o(t) = u_{ff}^{\rm st}(t) + u_{ff}^{\rm ust}(t).$$
 (17)

Then, $u_{ff}(t)$ is sampled into $u_{ff}[k]$ by a zero-order hold. No consideration of the zero-order hold causes a delay. This will be discussed in Section V-F and shown in Fig. 11.

IV. PREACTUATED MULTIRATE FEEDFORWARD CONTROL

We propose a multirate feedforward control with preactuation to design a stable inversion feedforward controller for plants with unstable intrinsic and discretization zeros. This method solves the unstable zero inversion problem in three steps. First, we calculate the controllable canonical form realization for the nominal plant. Second, the stable inversion for the unstable intrinsic zeros is calculated using a time- and imaginary-axis reversal in a continuous-time domain. Then, stable inversion for unstable discretization zeros is calculated using a multirate feedforward proposed in [27]. A block diagram of the preactuated multirate feedforward control is shown in Fig. 5.

A. Controllable Canonical Form Realization

We realize the nominal plant shown in (3) as the controllable canonical form

$$\boldsymbol{x}(t) = \begin{bmatrix} x_{1}(t) & x_{2}(t) & \cdots & x_{n}(t) \end{bmatrix}^{\mathrm{T}} \\ \boldsymbol{A}_{c} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ & & & \ddots & \\ -a_{0} & -a_{1} & -a_{2} & \cdots & -a_{n-1} \end{bmatrix}$$
(18)
$$\boldsymbol{b}_{c} = \begin{bmatrix} 0 & 0 & \cdots & b_{0} \end{bmatrix}^{\mathrm{T}} \\ \boldsymbol{c}_{c} = \begin{bmatrix} 1 & \frac{b_{1}}{b_{0}} & \cdots & \frac{b_{m}}{b_{0}} & 0 & \cdots & 0 \end{bmatrix}.$$

The objective to use the controllable canonical form is to use the differential relationship of the state vector, as shown in (20).

B. State Trajectory x_d Generation

First, a desired state trajectory x_d for the multirate feedforward is generated. The state trajectory x_d is defined as

$$\boldsymbol{x}_d(t) = \begin{bmatrix} x_{1d}(t) & x_{2d}(t) & \cdots & x_{nd}(t) \end{bmatrix}^{\mathrm{T}}.$$
 (19)

Owing to the controllable canonical form realization [see (18)], we have

$$\boldsymbol{x}_{d}(t) = \begin{bmatrix} x_{1d}(t) & \rho x_{1d}(t) & \cdots & \rho^{n-1} x_{1d}(t) \end{bmatrix}^{\mathrm{T}}.$$
 (20)

According to (3), to track the reference position trajectory r(t), the desired state trajectory $x_d(t)$ should satisfy

$$r(t) = c_{c} x_{d}(t)$$

$$= \begin{bmatrix} 1 & \frac{b_{1}}{b_{0}} & \cdots & \frac{b_{m}}{b_{0}} & 0 & \cdots & 0 \end{bmatrix} \begin{bmatrix} x_{1d}(t) \\ \rho x_{1d}(t) \\ \rho^{2} x_{1d}(t) \\ \vdots \\ \rho^{n-1} x_{1d}(t) \end{bmatrix}.$$
(21)

From (2) and (21), $x_{1d}(t)$ is obtained as

$$x_{1d}(t) = \frac{1}{B(\rho)}r(t).$$
 (22)

Therefore, the entire vector $\boldsymbol{x}_d(t)$ is obtained as

$$\boldsymbol{x}_d(t) = \frac{1}{B(\rho)} \boldsymbol{r}(t) \tag{23}$$

where

$$\boldsymbol{r}(t) = \begin{bmatrix} r_1(t) & r_2(t) & \cdots & r_n(t) \end{bmatrix}^{\mathrm{T}}$$
$$= \begin{bmatrix} 1 & \rho & \cdots & \rho^{n-1} \end{bmatrix}^{\mathrm{T}} \boldsymbol{r}(t).$$
 (24)

However, (23) has unstable poles when the plant $P_n(s)$ has unstable zeros. To prevent the diversion of the state trajectory $\boldsymbol{x}_d(t)$, the stable–unstable decomposition and time-axis reversal techniques are applied.

1) Step 1. Stable–unstable decomposition: Stable–unstable composition is performed as same as the CPMI method shown in Section III-B1

$$B(s)^{-1} = \frac{b_0}{b_m s^m + b_{m-1} s^{m-1} + \dots + b_0}$$
(25)

$$=F^{\rm st}(s) + F^{\rm ust}(s) \tag{26}$$

$$f^{\mathrm{st}}(t) = \mathcal{L}^{-1}\left[F^{\mathrm{st}}(s)\right], \bar{f}^{\mathrm{ust}}(t) = \mathcal{L}^{-1}\left[F^{\mathrm{ust}}(-s)\right].$$
(27)

2) Step 2. Stable part state trajectory generation: The desired state trajectory $\boldsymbol{x}_{d}^{\text{st}}(t)$ for the stable part is forwardly generated as follows:

$$\boldsymbol{x}_{d}^{\mathrm{st}}(t) = \begin{bmatrix} x_{1d}^{\mathrm{st}}(t) & x_{2d}^{\mathrm{st}}(t) & \cdots & x_{nd}^{\mathrm{st}}(t) \end{bmatrix}^{\mathrm{T}}$$
$$= \int_{-\infty}^{t} f^{\mathrm{st}}(t-\tau)\boldsymbol{r}(\tau)d\tau.$$
(28)

Equation (28) can be written as

$$\boldsymbol{x}_{d}^{\mathrm{st}}(t) = \int_{0}^{t} f^{\mathrm{st}}(t-\tau)\boldsymbol{r}(\tau)d\tau$$
(29)

assuming $\mathbf{r}(t) = \mathbf{0}$ when t < 0.

3) Step 3. Unstable part state trajectory generation: The desired state trajectory $x_d^{ust}(t)$ for the unstable part is generated by

$$\boldsymbol{x}_{d}^{\text{ust}}(t) = \begin{bmatrix} x_{1d}^{\text{ust}}(t) & x_{2d}^{\text{ust}}(t) & \cdots & x_{nd}^{\text{ust}}(t) \end{bmatrix}^{\text{T}} \\ = \int_{-\infty}^{\bar{t}} \bar{f}^{\text{ust}}(\bar{t} - \bar{\tau}) \boldsymbol{r}(-\bar{\tau}) d\bar{\tau} \Big|_{\bar{t}=-t}.$$
(30)

 $\boldsymbol{x}_{d}^{\text{ust}}(t)$ is calculated as follows: first, a convolution of the timereversed reference position trajectory $\boldsymbol{r}(-\bar{t})$ and the stable signal $\bar{f}^{\text{ust}}(\bar{t})$ is calculated. Next, the time axis is reversed. The mathematical proof is provided in [34].

4) Step 4. State trajectory generation: The state trajectory $x_d(t)$ is obtained by

$$\boldsymbol{x}_d(t) = \boldsymbol{x}_d^{\rm st}(t) + \boldsymbol{x}_d^{\rm ust}(t). \tag{31}$$

C. Feedforward Input Generation From the State Trajectory

The effect of unstable discretization zeros can be avoided using the multirate feedforward control [27]. Fig. 5 shows that there are three time periods T_y , T_u , and T_r denoting the periods for y(t), u(t), and r(t), respectively. These periods are set as $T_r = nT_u = nT_y$.

The multirate system of (4) is given as

$$x[i+1] = Ax[i] + Bu[i], y[i] = cx[i]$$
 (32)

where

by calculating the state transition from $t = iT_r = kT_u$ to $t = (i+1)T_r = (k+n)T_u$. Here, the input vector u[i] is defined in the lifting form

$$\boldsymbol{u}[i] = \begin{bmatrix} u_1[i] & u_2[i] & \cdots & u_n[i] \end{bmatrix}^{\mathrm{T}} \\ = \begin{bmatrix} u(kT_u) & u((k+1)T_u) & \cdots & u((k+n-1)T_u) \end{bmatrix}^{\mathrm{T}}.$$
(34)

According to (32), the feedforward input $u_o[i]$ is obtained from the previewed state trajectory $x_d[i+1]$ as follows:

$$u_o[i] = B^{-1}(I - z^{-1}A)x_d[i+1]$$
(35)

where $z = e^{sT_r}$.

V. EXPERIMENTAL VALIDATION

A. Experimental Setup

The experimental setup used to verify the proposed theory is shown in Fig. 6. It is an air-guided single-degree-offreedom flexible stage driven by a set of linear motors. The position of the table and the driven carriage are measured by two linear encoders with 1-nm precision. By using interior or exterior division, we can measure any vertical imaginary position. In this paper, the height of the measurement point is set as $L_m = 0.300$ m by exterior division to have continuous-time unstable zeros. The applied force, i.e., current, is measured through



Fig. 6. Experimental high-precision positioning stage and its frequency response function (FRF) data and eighth-order model ($L_m = 0.300$ [m]). (a) High-precision positioning stage. (b) FRF data and model of (a).



Fig. 7. Pole-zero map of the identified model shown in Fig. 6. (a) P_c . (b) P_s .

an inductive current sensor with a 25-A range and quantized with a 14-bit A/D converter, i.e., 3.05-mA step.

B. System Identification

The dominant system dynamics of the setup are modeled through frequency-domain identification techniques; see [35]. First, the nonparametric frequency response is measured from open-loop experiments using periodic multisine excitations with quasi-logarithmic spacing to cover the wide frequency band of interest. Subsequently, an eighth-order parametric transfer function model $P_n(s)$ is estimated iteratively using a Levenberg– Marquardt method with a maximum likelihood criterion. Some additional weighting is applied in the middle frequency range (10–400 Hz) to reflect the region in which high model accuracy is desired for feedforward design.

The measured frequency response and the estimated eighthorder model are shown in Fig. 6(b). The identified continuoustime pole-zero map is shown in Fig. 7(a), in which $P_n(s)$ contains continuous-time unstable zeros at s = +126, +878and stable zeros at $s = -1090, -120, -92.7 \pm 1.20j$. Additionally, the pole-zero map of the discretized transfer function with $T_u = 400 \ \mu s$ is shown in Fig. 7(b), in which $P_s[z_s]$ contains an unstable discretization zero.

C. Conditions

The setup has a current controller as inner loop, which has a 1-kHz bandwidth, with a 12.5-kHz sampling and a position controller as outer loop with 2.5-kHz sampling. The position feedforward controllers are designed with $T_u = 400 \ \mu s$ sampling time. The position feedback controller is designed



Fig. 8. Feedback control performance. The feedback controller is designed as a PID controller and a second-order shaping filter. The designed gain and phase margins are 14.2 dB (at 10 Hz) and 37.2° (at 2.8 Hz), respectively. (a) Bode plot of the open-loop system. (b) Nyquist plot of the open-loop system.

as a proportional-integral-derivative (PID) controller and a second-order shaping filter. The feedback control performance is shown in Fig. 8, which shows that it is difficult to achieve high bandwidth for a plant with continuous-time unstable zeros in the low-frequency range. This indicates that the feedback controller cannot help the trajectory tracking performance. The block diagram shown in Fig. 2 is used for NPZI, ZPETC, and ZMETC methods. Note that in the configurations of Figs. 2, 4, and 5, without modeling error or disturbances, the output of the feedback controller $C_{\text{fb}}[z_s]$ is zero.

The target trajectory is given as a 0.05 second step reference interpolated by a 15th-order polynomial and is shown in, i.e., Fig. 9(a). As for the CMPI and the proposed methods, the control input is applied from t = -0.0428 s to preactuate the system. This time length is determined by the current sensor resolution. This is 5.38 times longer than the time constant of the dominant unstable zero in continuous-time domain. The effect regarding short time preactuation compared to the time constant of the unstable zeros is discussed in [36].

D. Simulation Results

Simulation results are shown in Fig. 9. Fig. 9(f) demonstrates that the proposed method can achieve perfect tracking without any undershoot or overshoot. In contrast, Fig. 9(a) and (e) shows that the NPZI, ZMETC, and ZPETC controllers create undershoot and/or overshoot. The feedback only (FB only) case shows a slow response, which demonstrates that a plant with a slow continuous unstable zero cannot rely on a feedback controller only for reference tracking. Fig. 9(g) shows that the CMPI method creates a similar current reference to the proposed method. However, due to the lack of considering the zero-order hold, the current reference is delayed compared to the proposal, and it results in the tracking error shown in Fig. 9(f). Note that these simulations contain no modeling error and/or disturbances; hence, the feedback current is zero for NPZI, ZPETC, ZMETC, CMPI, and the proposed method.

E. Experimental Results

The experimental results are shown in Fig. 10. The trend in the experimental results are in good agreement with the simulations



Fig. 9. Simulation with the eighth-order model shown in Fig. 6(b). Note that in (d) and (h), feedback current is zero for NPZI, ZPETC, ZMETC, CPMI, and the proposal due to no modeling error and disturbance assumption. (a) Position. (b) Tracking error. (c) Feedforward current. (d) Feedback current. (e) Position. (f) Tracking error. (g) Feedforward current. (h) Feedback current.



Fig. 10. Experiment with the stage shown in Fig. 6. The eighth-order nominal model is used for feedforward controller design. (a) Position. (b) Tracking error. (c) Feedforward current. (d) Feedback current. (e) Position. (f) Tracking error. (g) Feedforward current. (h) Feedback current.

shown in Fig. 9. From Fig. 10(e) and (g), during preactuation, the output position has almost no motion. After preactuation, the proposed method has almost no undershoot or overshoot. As summarized in Table I, the proposed method is experimentally validated.

Fig. 10(d) and (h) shows that, except the feedback only case, the feedback current references are quite small compared to the feedforward current references. This is because the nominal output $y_o[k]$, which is calculated by the feedforward current reference $u_o[k]$ and the nominal plant $P_n[z_s]$, almost matches the actual output y[k] due to the well-identified nominal model.

TABLE I MAXIMUM TRACKING ERROR (IN MICROMETERS) OF FIGS. 9 AND 10

	FB only	NPZI	ZMETC	ZPETC	CPMI	Proposal
Sim	1590	1020	1360	658	20.0	0.00475*
Exp	1580	1020	1300	631	72.3	41.1

*Intersample tracking error.

F. Observations of the Effect of Multirate Feedforward

As described above, the feedforward current reference is similar between the CPMI method and the proposed method,



Fig. 11. Error comparison between the proposed method and the shifted CPMI method. Dots are illustrated by every $T_r = nT_u$. It shows that the proposed method achieves the perfect tracking for $T_r = nT_u$, which is not achievable by the CPMI method (simulation). (a) Tracking error. (b) Tracking error (zoom). (c) Tracking error (zoom2).

as shown in Figs. 9(g) and 10(g). The difference of the two methods is the zero-order hold consideration by the multirate feedforward.

Fig. 11 shows the tracking error comparison between the proposed and shifted CPMI methods. It shows that the zero-order hold delay cannot be compensated by just shifting the current reference even by a noninteger sample shift. These results clearly show the importance of multirate feedforward control, which compensates for the zero-order hold delay. Fig. 11(c) shows that perfect tracking is achieved by the proposed method for every $T_r = nT_u$.

VI. CONCLUSION

In the discretized domain, there are two types of zeros: 1) the intrinsic zeros, which have counterparts in the continuoustime domain; and 2) the discretization zeros generated by discretization by the zero-order hold. The proposed preactuated multirate feedforward control deals with problems 1) and 2) independently. On one hand, the unstable intrinsic zeros are compensated by the preactuation. On the other hand, the unstable discretization zeros are compensated by the multirate feedforward control with preview. The multirate feedforward controller generates the feedforward input, which achieves perfect tracking for the designed state trajectory. The simulation results show that the zero-order hold delay cannot be compensated by just shifting the reference and underline the importance of multirate feedforward.

This study experimentally validates the proposed method using a high-precision positioning stage with continuous-time unstable zeros. Additionally, this system has a discretization zero. Owing to a well-identified eighth-order model, the experimental results follow the simulations. The experimental result obtained with the proposed method strongly reduces the tracking error and achieves almost zero undershoot and overshoot.

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